## **CHAPTER**

# 

# **IGNITION**

(GE90-100 SERIES ENGINES)



## CHAPTER 74 IGNITION

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A = Added, R = Revised, D = Deleted, O = Overflow, C = Customer Originated Change

## 74-EFFECTIVE PAGES



YOU FIND A FAULT WITH AN AIRPLANE SYSTEM These are the possible types of faults:

- 1. EICAS Message
- 2. Observed Fault
- 3. Cabin Fault
- 4. Non-Correlated
  Maintenance Message

USE THE MAT TO GET MORE INFORMATION

If you have an EICAS message, go to the MAT to find its fault code and the corresponding maintenance message numbers.

For details, see Figure 2 →

GO TO THE FAULT ISOLATION TASK IN THE FIM

Use the fault code or description to find the task in the FIM. There is a numerical list of fault codes in each chapter. There are lists of fault descriptions at the front of the FIM.

For details, see Figure 3 -

FOLLOW THE STEPS OF THE FAULT ISOLATION TASK

The fault isolation task explains how to find the cause of the fault. When the task says "You corrected the fault" you know that the fault is gone.

For details, see Figure 4 ──►

E84424 S0000132469\_V1

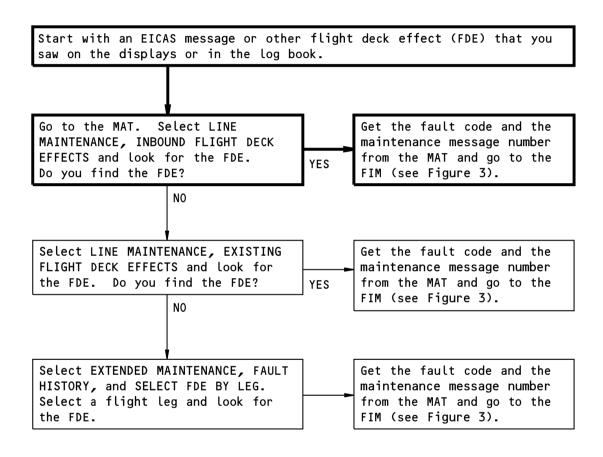
Basic Fault Isolation Process Figure 1

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NOTE: The bold lines show the most common path.

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## Getting Fault Information from the MAT Figure 2

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#### IF YOU HAVE:

#### THEN DO THIS TO FIND THE TASK IN THE FIM:

#### FAULT CODE

with maintenance
message number
(if applicable)

- The first two digits of the fault code are the FIM chapter that you need. Go to the Fault Code Index in that chapter and find the fault code. If the fault code starts with a letter, then go to the Cabin Fault Code Index at the front of the FIM.
- 2. Find the maintenance message number (if there is one) to the right of the fault code.
- 3. Find the task number on the same line as the maintenance message number. Go to the task in the FIM and do the steps in the task (see Figure 4).

#### EICAS MESSAGE TEXT

with no fault code

 Go to the MAT. Find the fault code and the correlated maintenance message number (see Figure 2). Then do the FAULT CODE procedure above.

## OBSERVED FAULT DESCRIPTION

or cabin fault description

- 1. Go to the Observed Fault List or Cabin Fault List at the front of the FIM and find the best description for the fault.
- 2. Find the task number on the same line as the fault description. Go to the task in the FIM and do the steps of the task (see Figure 4).

#### The first two digits of the maintenance message number are the FIM chapter you need. Go to the Maintenance Message Index in that chapter and find the maintenance message number.

2. Find the task number on the same line as the maintenance message number. Go to the task in the FIM and do the steps in the task (see Figure 4).

MAINTENANCE MESSAGE NUMBER

with no correlated EICAS message

NOTE: When you troubleshoot Non-correlated Maintenance Messages, you must plan for sufficient resources and the necessary time and parts to perform the applicable FIM Procedure from Start to Finish (or until the fault goes away). If you do not complete the procedure and clear the fault, in some cases additional faults can be set which could possibly cause unscheduled delays and/or Airplane-on-Ground (AOG) conditions.

E84427 S0000132476\_V2

Finding the Fault Isolation Task in the FIM Figure 3

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#### ASSUMED CONDITIONS AT START OF TASK

- External electrical power is ON
- Hydraulic power and pneumatic power are OFF
- Engines are shut down
- No equipment in the system is deactivated

#### INITIAL EVALUATION PARAGRAPH

- The Initial Evaluation paragraph at the start of the task helps you determine whether you can detect the fault right now.
- If you cannot detect the fault right now, then the task cannot isolate the fault and the Initial Evaluation paragraph will say that there was an intermittent fault.
- If you have an intermittent fault, you must use your judgement (and follow your airline's policy) to decide which components to replace. Then monitor the airplane to see if the fault happens again on subsequent flights.

#### FAULT ISOLATION STEPS

- The FIM task steps are presented in a specified order.
  "The If... then" statements will guide you along a logical path.
  But if you do not plan to follow the FIM task exactly, make sure
  that you read it before you start to isolate the fault. Some
  FIM procedures start with important steps that have an effect on
  the other steps in the procedure.
- When you are at the endpoint of the path, the step says "You corrected the fault." Complete the step and exit the procedure.
- The Recommended Maintenance Action that shows on the MAT for the maintenance message gives a list of possible causes in order by probability of failure. In the FIM procedure, the possible causes can be in a different order from the MAT.

#### **WIRING CHECKS**

When a step says "Do a wiring check", do these three types of electrical checks for the specified contacts (pins):

- continuity from contact to contact
- shorts between the contacts
- shorts from each contact to ground

E84428 S0000132477\_V3

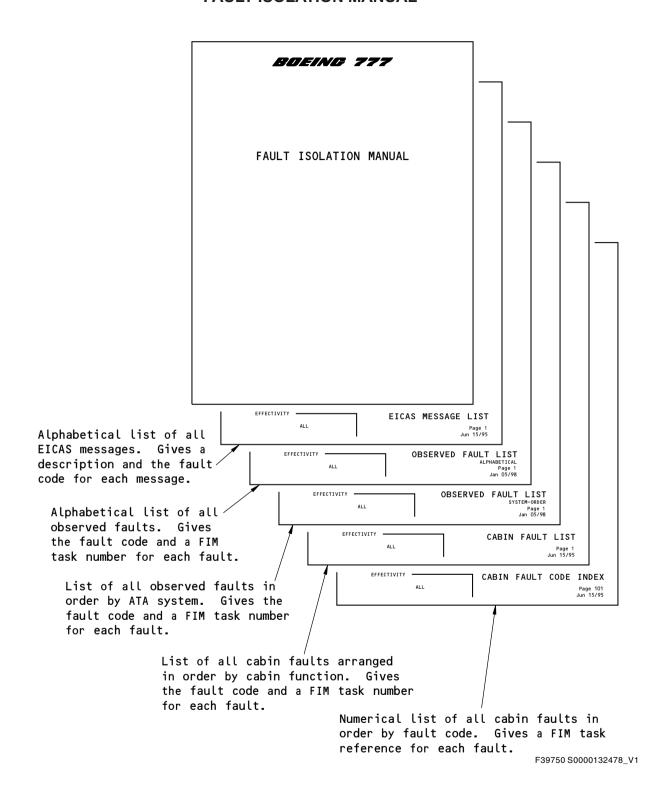
Doing the Fault Isolation Task Figure 4

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Subjects at Front of FIM Figure 5

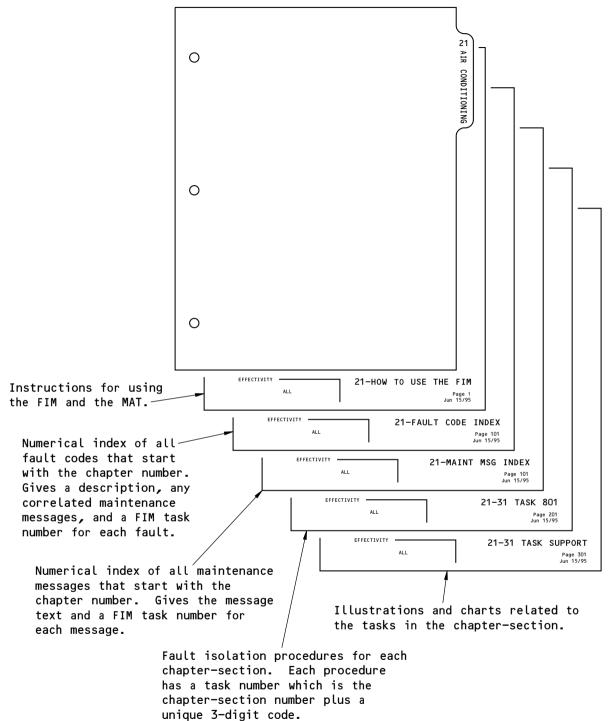
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Subjects in Each FIM Chapter Figure 6

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FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
742 001 51	ENG IGNITOR L1 (EICAS STATUS)		
	Before you do any task listed here, see FIM 74-98 TASK 802.		
	<b>NOTE:</b> AIMS CAN LATCH THIS MESSAGE.		
		24-11786	24-51 TASK 833
		24-13817	24-33 TASK 818
		74-33921	74-21 TASK 807
		74-33941	74-21 TASK 812
742 001 52	ENG IGNITOR R1 (EICAS STATUS)		
	Before you do any task listed here, see FIM 74-98 TASK 802.		
	<b>NOTE</b> : AIMS CAN LATCH THIS MESSAGE.	04.4040=	04.54.54.04.040
		24-12127	24-51 TASK 842
		24-13817	24-33 TASK 818
		74-33922	74-21 TASK 807
		74-33942	74-21 TASK 813
742 002 51	ENG IGNITOR L2 (EICAS STATUS)		
	Before you do any task listed here, see FIM 74-98 TASK 802.		
	<b>NOTE</b> : AIMS CAN LATCH THIS MESSAGE.	04.4040=	04.54.74.014.040
		24-12127	24-51 TASK 842
		24-13817	24-33 TASK 818
		74-33931	74-21 TASK 808
		74-33951	74-21 TASK 815
742 002 52	ENG IGNITOR R2 (EICAS STATUS)		
	Before you do any task listed here, see FIM 74-98 TASK 802.		
	<b>NOTE</b> : AIMS CAN LATCH THIS MESSAGE.	04.44700	04.54.74.014.000
		24-11786	24-51 TASK 833
		24-13817	24-33 TASK 818
		74-33932	74-21 TASK 808
		74-33952	74-21 TASK 816
743 499 50	CON IGNITION ON L+R (EICAS MEMO)	23-41420	23-93 TASK 801
		23-41530	23-93 TASK 801
		23-41540	23-93 TASK 801
		23-41550	23-93 TASK 801
		23-41560	23-93 TASK 801
		23-41870	23-93 TASK 803

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FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
743 499 50	CON IGNITION ON L+R (EICAS MEMO)	(continued)	
		23-41900	23-93 TASK 803
		23-41910	23-93 TASK 803
		23-42420	23-93 TASK 802
		23-42530	23-93 TASK 802
		23-42540	23-93 TASK 802
		23-42550	23-93 TASK 802
		23-42560	23-93 TASK 802
		23-42870	23-93 TASK 804
		23-42900	23-93 TASK 804
		23-42910	23-93 TASK 804
		23-43420	73-29 TASK 824
		23-43530	28-22 TASK 810
		23-43540	28-22 TASK 812
		23-43550	30-21 TASK 809
		23-43560	74-21 TASK 805
1		23-43870	73-29 TASK 825
		23-43900	30-21 TASK 810
I		23-43910	74-21 TASK 806
		23-45010	23-93 TASK 881
		23-46010	23-93 TASK 806
		23-47010	23-93 TASK 801
		23-47020	23-93 TASK 802
		23-48010	23-93 TASK 803
		23-48020	23-93 TASK 804
		23-81001	23-91 TASK 801
		23-81002	23-91 TASK 802
		23-81003	23-91 TASK 803
		23-81004	23-91 TASK 804
		23-81005	23-91 TASK 805
		23-81006	23-91 TASK 806
		23-81007	23-91 TASK 807
		23-91010	23-93 TASK 816
		23-91020	23-93 TASK 878
		23-92010	23-93 TASK 879
		23-92020	23-93 TASK 880

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FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
743 499 50	CON IGNITION ON L+R (EICAS MEMO)	(continued)	
		24-11787	24-32 TASK 817
		24-12051	24-32 TASK 819
		24-12056	24-32 TASK 820
		24-12124	24-51 TASK 840
		24-12127	24-51 TASK 842
		24-13801	24-61 TASK 805
		24-13815	24-61 TASK 806
		24-13816	24-32 TASK 831
		24-13817	24-33 TASK 818
		24-13818	24-53 TASK 807
		27-20251	27-51 TASK 863
		27-20252	27-52 TASK 809
		27-20261	27-51 TASK 809
		27-20262	27-52 TASK 810
		27-20271	27-51 TASK 810
		27-20272	27-52 TASK 811
		27-20891	27-51 TASK 823
		27-20892	27-52 TASK 826
		27-21241	27-51 TASK 838
		27-21242	27-52 TASK 832
		27-28801	27-51 TASK 841
		27-28802	27-52 TASK 835
		27-28851	27-51 TASK 869
		27-28852	27-52 TASK 837
		31-50007	31-44 TASK 821
		31-50008	31-44 TASK 822
		31-50017	31-44 TASK 823
		31-50018	31-44 TASK 824
		31-50027	31-44 TASK 825
		31-50028	31-44 TASK 826
		31-50201	31-43 TASK 852
		31-50202	31-43 TASK 853
		31-50417	31-44 TASK 884
		31-50418	31-44 TASK 885
		31-50427	31-44 TASK 886

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FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
743 499 50	CON IGNITION ON L+R (EICAS MEMO)	(continued)	
		31-50428	31-44 TASK 887
		31-50437	31-44 TASK 888
		31-50438	31-44 TASK 889
		31-50447	31-44 TASK 890
		31-50448	31-44 TASK 891
		31-52013	31-43 TASK 866
		31-52014	31-43 TASK 867
		31-52101	31-41 TASK 817
		31-52112	31-41 TASK 820
		31-52121	31-41 TASK 821
		31-52122	31-41 TASK 822
		31-52601	31-41 TASK 832
		31-52602	31-41 TASK 833
		31-54013	31-43 TASK 868
		31-54014	31-43 TASK 869
		31-54101	31-41 TASK 834
		31-54102	31-41 TASK 835
		31-54121	31-41 TASK 838
		31-54122	31-41 TASK 839
		31-54601	31-41 TASK 840
		31-54602	31-41 TASK 841
		31-56013	31-43 TASK 870
		31-56014	31-43 TASK 871
		31-56102	31-41 TASK 845
		31-56111	31-41 TASK 846
		31-56121	31-41 TASK 848
		31-56122	31-41 TASK 849
		31-56601	31-41 TASK 842
		31-56602	31-41 TASK 843
		31-58013	31-43 TASK 872
		31-58014	31-43 TASK 873
		31-58101	31-41 TASK 852
		31-58102	31-41 TASK 853
		31-58121	31-41 TASK 856
		31-58122	31-41 TASK 857

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FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
743 499 50	CON IGNITION ON L+R (EICAS MEMO)	(continued)	
		31-58601	31-41 TASK 992
		31-58602	31-41 TASK 993
		31-61031	31-41 TASK 862
		31-61032	31-41 TASK 891
		31-61041	31-43 TASK 835
		31-61042	31-43 TASK 836
		31-61111	31-43 TASK 856
		31-61112	31-43 TASK 856
		31-61113	31-43 TASK 856
		31-61114	31-43 TASK 856
		31-61115	31-43 TASK 856
		31-61116	31-43 TASK 856
		31-61117	31-43 TASK 856
		31-61118	31-43 TASK 856
		31-62031	31-43 TASK 846
		31-62032	31-43 TASK 847
		31-62041	31-43 TASK 848
		31-62042	31-43 TASK 849
		31-69501	31-42 TASK 852
		31-69502	31-42 TASK 852
		31-69533	31-42 TASK 854
		31-69534	31-42 TASK 854
		31-69535	31-42 TASK 854
		31-69536	31-42 TASK 854
		73-03021	73-26 TASK 801
		73-03022	73-26 TASK 802
		73-03051	73-26 TASK 805
		73-03052	73-26 TASK 806
		73-03101	73-28 TASK 810
		73-03102	73-28 TASK 811
		73-03111	73-28 TASK 814
		73-03112	73-28 TASK 815
		73-03221	73-26 TASK 833
		73-03222	73-26 TASK 834
		73-03251	73-26 TASK 845

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FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
743 499 50	CON IGNITION ON L+R (EICAS MEMO)	(continued)	
		73-03252	73-26 TASK 846
		73-04011	73-27 TASK 801
		73-04012	73-27 TASK 802
		73-04031	73-27 TASK 801
		73-04032	73-27 TASK 802
		73-04131	73-28 TASK 830
		73-04132	73-28 TASK 830
		73-04141	73-28 TASK 830
		73-04142	73-28 TASK 830
		73-04151	73-28 TASK 830
		73-04152	73-28 TASK 830
		73-09401	73-27 TASK 808
		73-09403	73-27 TASK 808
		73-09602	73-27 TASK 809
		73-09604	73-27 TASK 809
		73-15241	73-26 TASK 837
		73-15242	73-26 TASK 838
		73-15251	73-26 TASK 839
		73-15252	73-26 TASK 840
		73-25261	73-26 TASK 841
		73-25262	73-26 TASK 842
		73-25271	73-26 TASK 843
		73-25272	73-26 TASK 844
		73-35301	73-28 TASK 847
		73-35302	73-28 TASK 848
		73-35521	73-29 TASK 820
		73-35522	73-29 TASK 821
		73-35581	73-34 TASK 812
		73-35582	73-34 TASK 812
743 499 51	CON IGNITION ON L (EICAS MEMO)	23-41420	23-93 TASK 801
		23-41530	23-93 TASK 801
		23-41540	23-93 TASK 801
		23-41550	23-93 TASK 801
		23-41560	23-93 TASK 801
		23-42420	23-93 TASK 802

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FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
743 499 51	CON IGNITION ON L (EICAS MEMO)	(continued)	
		23-42530	23-93 TASK 802
		23-42540	23-93 TASK 802
		23-42550	23-93 TASK 802
		23-42560	23-93 TASK 802
		23-43420	73-29 TASK 824
		23-43530	28-22 TASK 810
		23-43540	28-22 TASK 812
		23-43550	30-21 TASK 809
		23-43560	74-21 TASK 805
		23-45010	23-93 TASK 881
		23-46010	23-93 TASK 806
		23-47010	23-93 TASK 801
		23-47020	23-93 TASK 802
		23-81001	23-91 TASK 801
		23-81002	23-91 TASK 802
		23-81004	23-91 TASK 804
		23-81005	23-91 TASK 805
		23-81006	23-91 TASK 806
		23-91010	23-93 TASK 816
		23-91020	23-93 TASK 878
		24-11787	24-32 TASK 817
		24-12051	24-32 TASK 819
		24-12056	24-32 TASK 820
		24-12124	24-51 TASK 840
		24-12127	24-51 TASK 842
		24-13801	24-61 TASK 805
		24-13815	24-61 TASK 806
		24-13816	24-32 TASK 831
		24-13817	24-33 TASK 818
		24-13818	24-53 TASK 807
		27-20251	27-51 TASK 863
		27-20252	27-52 TASK 809
		27-20261	27-51 TASK 809
		27-20262	27-52 TASK 810
		27-20271	27-51 TASK 810

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FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
743 499 51	CON IGNITION ON L (EICAS MEMO)	(continued)	
		27-20272	27-52 TASK 811
		27-20891	27-51 TASK 823
		27-20892	27-52 TASK 826
		27-21241	27-51 TASK 838
		27-21242	27-52 TASK 832
		27-28802	27-52 TASK 835
		27-28852	27-52 TASK 837
		31-50007	31-44 TASK 821
		31-50008	31-44 TASK 822
		31-50017	31-44 TASK 823
		31-50018	31-44 TASK 824
		31-50027	31-44 TASK 825
		31-50028	31-44 TASK 826
		31-50201	31-43 TASK 852
		31-50202	31-43 TASK 853
		31-50427	31-44 TASK 886
		31-50428	31-44 TASK 887
		31-50437	31-44 TASK 888
		31-50438	31-44 TASK 889
		31-50447	31-44 TASK 890
		31-50448	31-44 TASK 891
		31-56013	31-43 TASK 870
		31-56014	31-43 TASK 871
		31-56102	31-41 TASK 845
		31-56111	31-41 TASK 846
		31-56121	31-41 TASK 848
		31-56122	31-41 TASK 849
		31-56601	31-41 TASK 842
		31-56602	31-41 TASK 843
		31-58013	31-43 TASK 872
		31-58014	31-43 TASK 873
		31-58101	31-41 TASK 852
		31-58102	31-41 TASK 853
		31-58121	31-41 TASK 856
		31-58122	31-41 TASK 857

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FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
743 499 51	CON IGNITION ON L (EICAS MEMO)	(continued)	
		31-58601	31-41 TASK 992
		31-58602	31-41 TASK 993
		31-61031	31-41 TASK 862
		31-61032	31-41 TASK 891
		31-61041	31-43 TASK 835
		31-61042	31-43 TASK 836
		31-61111	31-43 TASK 856
		31-61112	31-43 TASK 856
		31-61113	31-43 TASK 856
		31-61114	31-43 TASK 856
		31-69501	31-42 TASK 852
		31-69502	31-42 TASK 852
		31-69535	31-42 TASK 854
		31-69536	31-42 TASK 854
		73-03021	73-26 TASK 801
		73-03051	73-26 TASK 805
		73-03101	73-28 TASK 810
		73-03111	73-28 TASK 814
		73-03221	73-26 TASK 833
		73-03251	73-26 TASK 845
		73-04011	73-27 TASK 801
		73-04031	73-27 TASK 801
		73-04131	73-28 TASK 830
		73-04141	73-28 TASK 830
		73-04151	73-28 TASK 830
		73-09401	73-27 TASK 808
		73-09403	73-27 TASK 808
		73-15241	73-26 TASK 837
		73-15251	73-26 TASK 839
		73-25261	73-26 TASK 841
		73-25271	73-26 TASK 843
		73-35301	73-28 TASK 847
		73-35521	73-29 TASK 820
		73-35581	73-34 TASK 812

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	FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
	743 499 52	CON IGNITION ON R (EICAS MEMO)	23-41870	23-93 TASK 803
			23-41900	23-93 TASK 803
			23-41910	23-93 TASK 803
			23-42870	23-93 TASK 804
			23-42900	23-93 TASK 804
			23-42910	23-93 TASK 804
I			23-43870	73-29 TASK 825
			23-43900	30-21 TASK 810
I			23-43910	74-21 TASK 806
			23-45010	23-93 TASK 881
			23-46010	23-93 TASK 806
			23-48010	23-93 TASK 803
			23-48020	23-93 TASK 804
			23-81002	23-91 TASK 802
			23-81003	23-91 TASK 803
			23-81004	23-91 TASK 804
			23-81005	23-91 TASK 805
			23-81006	23-91 TASK 806
			23-81007	23-91 TASK 807
			23-92010	23-93 TASK 879
			23-92020	23-93 TASK 880
			24-11787	24-32 TASK 817
			24-12051	24-32 TASK 819
			24-12056	24-32 TASK 820
			24-12124	24-51 TASK 840
			24-12127	24-51 TASK 842
			24-13801	24-61 TASK 805
			24-13815	24-61 TASK 806
			24-13816	24-32 TASK 831
			24-13817	24-33 TASK 818
			24-13818	24-53 TASK 807
			27-20251	27-51 TASK 863
			27-20252	27-52 TASK 809
			27-20261	27-51 TASK 809
			27-20262	27-52 TASK 810
			27-20271	27-51 TASK 810

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FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
743 499 52	CON IGNITION ON R (EICAS MEMO)	(continued)	
		27-20272	27-52 TASK 811
		27-20891	27-51 TASK 823
		27-20892	27-52 TASK 826
		27-21241	27-51 TASK 838
		27-21242	27-52 TASK 832
		27-28801	27-51 TASK 841
		27-28851	27-51 TASK 869
		31-50007	31-44 TASK 821
		31-50008	31-44 TASK 822
		31-50017	31-44 TASK 823
		31-50018	31-44 TASK 824
		31-50027	31-44 TASK 825
		31-50028	31-44 TASK 826
		31-50201	31-43 TASK 852
		31-50202	31-43 TASK 853
		31-50417	31-44 TASK 884
		31-50418	31-44 TASK 885
		31-50427	31-44 TASK 886
		31-50428	31-44 TASK 887
		31-50437	31-44 TASK 888
		31-50438	31-44 TASK 889
		31-50447	31-44 TASK 890
		31-50448	31-44 TASK 891
		31-52013	31-43 TASK 866
		31-52014	31-43 TASK 867
		31-52101	31-41 TASK 817
		31-52112	31-41 TASK 820
		31-52121	31-41 TASK 821
		31-52122	31-41 TASK 822
		31-52601	31-41 TASK 832
		31-52602	31-41 TASK 833
		31-54013	31-43 TASK 868
		31-54014	31-43 TASK 869
		31-54101	31-41 TASK 834
		31-54102	31-41 TASK 835

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FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
743 499 52	CON IGNITION ON R (EICAS MEMO)	(continued)	
		31-54121	31-41 TASK 838
		31-54122	31-41 TASK 839
		31-54601	31-41 TASK 840
		31-54602	31-41 TASK 841
		31-56013	31-43 TASK 870
		31-56014	31-43 TASK 871
		31-56121	31-41 TASK 848
		31-56122	31-41 TASK 849
		31-61115	31-43 TASK 856
		31-61116	31-43 TASK 856
		31-61117	31-43 TASK 856
		31-61118	31-43 TASK 856
		31-62031	31-43 TASK 846
		31-62032	31-43 TASK 847
		31-62041	31-43 TASK 848
		31-62042	31-43 TASK 849
		31-69501	31-42 TASK 852
		31-69502	31-42 TASK 852
		31-69533	31-42 TASK 854
		31-69534	31-42 TASK 854
		73-03022	73-26 TASK 802
		73-03052	73-26 TASK 806
		73-03102	73-28 TASK 811
		73-03112	73-28 TASK 815
		73-03222	73-26 TASK 834
		73-03252	73-26 TASK 846
		73-04012	73-27 TASK 802
		73-04032	73-27 TASK 802
		73-04132	73-28 TASK 830
		73-04142	73-28 TASK 830
		73-04152	73-28 TASK 830
		73-09602	73-27 TASK 809
		73-09604	73-27 TASK 809
		73-15242	73-26 TASK 838
		73-15252	73-26 TASK 840

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#### **GE90-100 SERIES ENGINES**



## 777-200/300 FAULT ISOLATION MANUAL

FAULT CODE	FAULT DESCRIPTION	MAINT MSG	GO TO FIM TASK
743 499 52	CON IGNITION ON R (EICAS MEMO)	(continued)	
		73-25262	73-26 TASK 842
		73-25272	73-26 TASK 844
		73-35302	73-28 TASK 848
		73-35522	73-29 TASK 821
		73-35582	73-34 TASK 812

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MAINT MESSAGE	MESSAGE TEXT	GO TO FIM TASK
74-33921*	Upper Ignition System (L Eng) is failed.	74-21 TASK 807
74-33922*	Upper Ignition System (R Eng) is failed.	74-21 TASK 807
74-33931*	Lower Ignition System (L Eng) is failed.	74-21 TASK 808
74-33932*	Lower Ignition System (R Eng) is failed.	74-21 TASK 808
74-33941*	Upper Ignition System (L Eng) power is not available.	74-21 TASK 812
74-33942*	Upper Ignition System (R Eng) power is not available.	74-21 TASK 813
74-33951*	Lower Ignition System (L Eng) power is not available.	74-21 TASK 815
74-33952*	Lower Ignition System (R Eng) power is not available.	74-21 TASK 816

<sup>\*</sup>If the MAT shows LATCHED for the correlated EICAS message, then you must erase the EICAS message after you complete the FIM task.

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#### 805. L Start/Ignition Selector Switch is Failed - Fault Isolation

#### A. Maintenance Messages

(1) This task is for maintenance message: 23-43560.

#### B. Fault Isolation Procedure

- (1) Replace the L START/IGNITION selector switch, S80100 (WDM 74-11-11).
  - (a) Do the repair confirmation procedure at the end of this task.
- (2) Look for a failure in the wire harness between the L START/IGNITION selector switch, S80100, and the left overhead panel card file (OPCF), M23117.
  - (a) Disconnect the electrical connectors, DS80100 from the L START/IGNITION selector switch, S80100, and DM23117K from the left OPCF, M23117.
  - (b) Look for wire to wire shorts, shorts to ground, and continuity in the circuit as shown below (WDM 74-11-11):

NOTE: When you look for shorts to ground, the insulation resistance must be more than 100K ohms.

DS80100	DM23117K	
40		A7
38		A8

- (c) If you find a problem in the wire harness, repair the wire harness.
- (d) Do the repair confirmation procedure at the end of this task.
- (3) Replace the left OPCF, M23117.

These are the tasks:

Overhead Panel Card File (OPCF) Removal, AMM TASK 23-93-02-000-802

Overhead Panel Card File (OPCF) Installation, AMM TASK 23-93-02-400-802.

(a) Do the repair confirmation procedure at the end of this task.

#### C. Repair Confirmation

- Select the Existing Fault Screen on the MAT.
- (2) If the MAT shows the maintenance message, then continue with the subsequent step of this fault isolation procedure.
- (3) If the MAT does not show the maintenance message, then you corrected the fault.



#### 806. R Start/Ignition Selector Switch is Failed - Fault Isolation

#### A. Maintenance Messages

(1) This task is for maintenance message: 23-43910.

#### B. Fault Isolation Procedure

- (1) Replace the R START/IGNITION selector switch, S80200 (WDM 74-11-21).
  - (a) Do the repair confirmation procedure at the end of this task.
- (2) Look for a failure in the wire harness between the R START/IGNITION selector switch, S80200, and the right overhead panel card file (OPCF), M23217.
  - (a) Disconnect the electrical connectors, DS80200 from the R START/IGNITION selector switch, S80200, and DM23217J from the right OPCF, M23217.

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(b) Look for wire to wire shorts, shorts to ground, and continuity in the circuit as shown below (WDM 74-11-21):

NOTE: When you look for shorts to ground, the insulation resistance must be more than 100K ohms.

DS80200	DM23217J
40	B14
38	B15

- (c) If you find a problem in the wire harness, repair the wire harness.
- (d) Do the repair confirmation procedure at the end of this task.
- (3) Replace the right OPCF, M23217.

These are the tasks:

Overhead Panel Card File (OPCF) Removal, AMM TASK 23-93-02-000-802

Overhead Panel Card File (OPCF) Installation, AMM TASK 23-93-02-400-802.

(a) Do the repair confirmation procedure at the end of this task.

#### C. Repair Confirmation

- (1) Select the Existing Fault Screen on the MAT.
- (2) If the MAT shows the maintenance message, then continue with the subsequent step of this fault isolation procedure.
- (3) If the MAT does not show the maintenance message, then you corrected the fault.



#### 807. Ignition System #1 (Upper) Failed - Fault Isolation

#### A. Maintenance Messages

(1) This task is for maintenance messages: 74-33921, 74-33922.

#### B. Description

(1) The engine did not light-off when using ignition system #1 (upper) during a start sequence. Power is available from the aircraft for exciter power. This fault can be reported whenever the EEC is powered.

NOTE: This maintenance message may occur during a ground start in extremely cold conditions. If this message occurs during an attempted ground start after a cold soak condition, repeat the start attempt. If the message occurs after the second attempted start, do the fault isolation procedure below.

#### C. Fault Isolation Procedure

- (1) Do the procedure to examine the ignitor, and look for damage to the ceramic insulation and tip of the ignitor (AMM TASK 74-21-02-200-801-H01).
  - (a) If you see damage that is not in the limits, replace the system 1 (upper) ignitor.

These are the tasks:

Igniter (System 1 or System 2) Removal, AMM TASK 74-21-02-000-801-H01 Igniter (System 1 or System 2) Installation, AMM TASK 74-21-02-400-801-H01.

- Do the repair confirmation procedure at the end of this task.
- (b) If the ignitor inspection is satisfactory, then continue.

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(2) Make sure the applicable EEC MAINT POWER switch is in the NORM position.



REMOVE ELECTRICAL POWER FROM THE EEC BEFORE YOU DISCONNECT THE ELECTRICAL CONNECTORS. IF YOU DO NOT OBEY THIS INSTRUCTION, DAMAGE TO THE EEC CAN OCCUR.

(3) Disconnect connector DM74001 from the ignition exciter, M74001.



MAKE SURE YOU DISCONNECT CONNECTOR DM73003B FROM THE EEC (FADEC). IF YOU DO NOT, IGNITOR 2 COULD FIRE WHEN YOU APPLY POWER, AND CAUSE INJURIES TO PERSONS AND DAMAGE TO EQUIPMENT.

(4) Disconnect connector DM73003B from the EEC (FADEC), M73003.

NOTE: You do this step to disable the other half of the ignition system while you do the MAT initiated ignition system tests.

- (5) Do this task: Test No. 17 Engine Ignition System Test (MAT Initiated Test), AMM TASK 71-00-00-700-816-H01.
  - (a) Select the ignitor circuit #1.



BE CAREFUL. 115 VAC CAN CAUSE SERIOUS INJURY TO PERSONS.

- (6) Do a check for 115 VAC between pins A and B on connector DM74001.
- (7) If there is not 115 VAC between pins A and B, do these steps:
  - (a) Stop the ignitor test on the MAT.



REMOVE ELECTRICAL POWER FROM THE EEC BEFORE YOU DISCONNECT THE ELECTRICAL CONNECTORS. IF YOU DO NOT OBEY THIS INSTRUCTION, DAMAGE TO THE EEC CAN OCCUR.

- (b) Disconnect connector DM73003A from the EEC (FADEC), M73003.
- (c) Attach a jumper from pin A to pin B on connector DM74001.
- (d) Do a check for continuity from pins G and H on connector DM73003A.
- (e) If there is not continuity between pins G and H, connect connector DM73003B to the EEC (FADEC), M73003, and replace the applicable wiring harness.

These are the tasks:

EEC (FADEC) Electrical Harness Removal, AMM TASK 73-22-01-000-801-H01 EEC (FADEC) Electrical Harness Installation, AMM TASK 73-22-01-400-801-H01.

- 1) Do the repair confirmation procedure at the end of this task.
- (f) If there is continuity, or if the fault continues, then remove the jumper from pins A and B on connector DM74001 and measure the resistance between these pins on connector DM73003A and the electrical harness sheathing (ground):
  - 1) Pin G to GND, specified resistance >100K ohms

ARO ALL



- 2) Pin H to GND, specified resistance >100K ohms.
- (g) If the resistance is not in the specified range for each pair of pins, replace the applicable wiring harness.

These are the tasks:

EEC (FADEC) Electrical Harness Removal, AMM TASK 73-22-01-000-801-H01 EEC (FADEC) Electrical Harness Installation, AMM TASK 73-22-01-400-801-H01.

(h) If the resistance is in the range specified for each pair of pins, replace the EEC (FADEC), M73003.

These are the tasks:

EEC (FADEC) Removal, AMM TASK 73-21-15-000-801-H01 EEC (FADEC) Installation, AMM TASK 73-21-15-400-801-H01.

- (i) Do these steps:
  - 1) Connect connector DM74001 to the ignition exciter, M74001.
  - 2) Connect connector DM73003B to the EEC (FADEC), M73003.
  - 3) Do the repair confirmation procedure at the end of this task.
- (8) If there is 115 VAC between pins A and B, replace the applicable ignitor.

These are the tasks:

Igniter (System 1 or System 2) Removal, AMM TASK 74-21-02-000-801-H01 Igniter (System 1 or System 2) Installation, AMM TASK 74-21-02-400-801-H01.

- (a) Do these steps:
  - 1) Connect connector DM74001 to the ignition exciter, M74001.
  - Connect connector DM73003B to the EEC (FADEC), M73003.
  - 3) Do the repair confirmation procedure at the end of this task.
- (9) Do an inspection of the combustion section of the engine (AMM TASK 72-00-00-290-804-H01).
- (10) If the fault continues, replace the ignition exciter.

These are the tasks:

Ignition Exciter (System 1 or System 2) Removal, AMM TASK 74-11-01-000-801-H01 Ignition Exciter (System 1 or System 2) Installation, AMM TASK 74-11-01-400-801-H01.

- (a) Do the repair confirmation procedure at the end of this task.
- (11) If the fault continues, replace the ignition lead.

These are the tasks:

Ignition Lead (System 1 or System 2) Installation, AMM TASK 74-21-01-400-801-H01 Ignition Lead (System 1 or System 2) Installation, AMM TASK 74-21-01-400-801-H01.

(12) If the fault continues, replace the fuel nozzle closest to the igniter.

These are the tasks:

Fuel Nozzle Removal, AMM TASK 73-11-05-000-801-H01 Fuel Nozzle Installation, AMM TASK 73-11-05-400-801-H01.

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#### D. Repair Confirmation

- Do this task: Test No. 17 Engine Ignition System Test (MAT Initiated Test), AMM TASK 71-00-00-700-816-H01.
  - (a) If the MAT shows NOT ACTIVE for the maintenance message (or if the message does not show), then you corrected the fault.
  - (b) If the MAT shows ACTIVE for the maintenance message, then continue with this fault isolation procedure at the subsequent step.



#### 808. Ignition System #2 (Lower) Failed - Fault Isolation

#### A. Maintenance Messages

(1) This task is for maintenance messages: 74-33931, 74-33932.

#### B. Description

(1) The engine did not light-off when using ignition system #2 (lower) during a start sequence. Power is available from the aircraft for exciter power. This fault can be reported whenever the EEC is powered.

NOTE: This maintenance message may occur during a ground start in extremely cold conditions. If this message occurs during an attempted ground start after a cold soak condition, repeat the start attempt. If the message occurs after the second attempted start, do the fault isolation procedure below.

#### C. Fault Isolation Procedure

- (1) Do the procedure to examine the ignitor, and look for damage to the ceramic insulation and tip of the ignitor (AMM TASK 74-21-02-200-801-H01).
  - (a) If you see damage that is not in the limits, replace the system 2 (lower) ignitor.

These are the tasks:

Igniter (System 1 or System 2) Removal, AMM TASK 74-21-02-000-801-H01 Igniter (System 1 or System 2) Installation, AMM TASK 74-21-02-400-801-H01.

- 1) Do the repair confirmation procedure at the end of this task.
- (b) If the ignitor is satisfactory, then continue.
- (2) Make sure the applicable EEC MAINT POWER switch is in the NORM position.



REMOVE ELECTRICAL POWER FROM THE EEC BEFORE YOU DISCONNECT THE ELECTRICAL CONNECTORS. IF YOU DO NOT OBEY THIS INSTRUCTION, DAMAGE TO THE EEC CAN OCCUR.

(3) Disconnect connector DM74002 from the ignition exciter, M74002.



MAKE SURE YOU DISCONNECT CONNECTOR DM73003A FROM THE EEC (FADEC). IF YOU DO NOT, IGNITOR 1 COULD FIRE WHEN YOU APPLY POWER, AND CAUSE INJURIES TO PERSONS AND DAMAGE TO EQUIPMENT.

(4) Disconnect connector DM73003A from the EEC (FADEC), M73003.

NOTE: You do this step to disable the other half of the ignition system while you do the MAT initiated tests.

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- (5) Do this task: Test No. 17 Engine Ignition System Test (MAT Initiated Test), AMM TASK 71-00-00-700-816-H01.
  - (a) Select the ignitor circuit #2.



BE CAREFUL. 115 VAC CAN CAUSE SERIOUS INJURY TO PERSONS.

#### WARNING

- (6) Do a check for 115 VAC between pins A and B on connector DM74002.
- (7) If there is not 115 VAC between pins A and B, do these steps:
  - (a) Stop the ignitor test on the MAT.



REMOVE ELECTRICAL POWER FROM THE EEC BEFORE YOU DISCONNECT THE ELECTRICAL CONNECTORS. IF YOU DO NOT OBEY THIS INSTRUCTION, DAMAGE TO THE EEC CAN OCCUR.

- (b) Disconnect connector DM73003B from the EEC (FADEC), M73003.
- (c) Attach a jumper wire from pin A to pin B on connector DM74002.
- (d) Do a check for continuity from pins G and H on connector DM73003B.
- (e) If there is not continuity between pins G and H, connect connector DM73003B to the EEC (FADEC), M73003, and replace the applicable wiring harness.

These are the tasks:

EEC (FADEC) Electrical Harness Removal, AMM TASK 73-22-01-000-801-H01 EEC (FADEC) Electrical Harness Installation, AMM TASK 73-22-01-400-801-H01.

- 1) Do the repair confirmation procedure at the end of this task.
- (f) If there is continuity, or if the fault continues, then remove the jumper from pins A and B on connector DM74002 and measure the resistance between these pins on connector DM73003B and the electrical harness sheathing (ground):
  - 1) Pin G to GND, specified resistance >100K ohms.
  - 2) Pin H to GND, specified resistance >100K ohms.
- (g) If the resistance is not in the specified range for each pair of pins, replace the applicable wiring harness. These are the tasks:

EEC (FADEC) Electrical Harness Removal, AMM TASK 73-22-01-000-801-H01

EEC (FADEC) Electrical Harness Installation, AMM TASK 73-22-01-400-801-H01.

(h) If the resistance is in the range specified for each pair of pins, replace the EEC (FADEC), M73003. These are the tasks:

EEC (FADEC) Removal, AMM TASK 73-21-15-000-801-H01

EEC (FADEC) Installation, AMM TASK 73-21-15-400-801-H01.

(i) Do these steps:

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- 1) Connect connector DM74002 to the ignition exciter M74002.
- Connect connector DM73003A to the EEC (FADEC), M73003.
- 3) Do the repair confirmation procedure at the end of this task.
- (8) If there is 115 VAC between pins A and B, replace the applicable ignitor.

— EFFECTIVITY — 7*A*-21 **7** 



These are the tasks:

Igniter (System 1 or System 2) Removal, AMM TASK 74-21-02-000-801-H01 Igniter (System 1 or System 2) Installation, AMM TASK 74-21-02-400-801-H01.

- (a) Do these steps:
  - 1) Connect connector DM74002 to the ignition exciter, M74002.
  - 2) Connect connector DM73003A to the EEC (FADEC), M73003.
  - 3) Do the repair confirmation procedure at the end of this task.
- (9) Do an inspection of the combustion section of the engine (AMM TASK 72-00-00-290-804-H01).
- (10) If the fault continues, replace the ignition exciter.

These are the tasks:

Ignition Exciter (System 1 or System 2) Removal, AMM TASK 74-11-01-000-801-H01 Ignition Exciter (System 1 or System 2) Installation, AMM TASK 74-11-01-400-801-H01.

- (a) Do the repair confirmation procedure at the end of this task.
- (11) If the fault continues, replace the ignition lead.

These are the tasks:

Ignition Lead (System 1 or System 2) Installation, AMM TASK 74-21-01-400-801-H01 Ignition Lead (System 1 or System 2) Installation, AMM TASK 74-21-01-400-801-H01.

#### D. Repair Confirmation

- (1) Do this task: Test No. 17 Engine Ignition System Test (MAT Initiated Test), AMM TASK 71-00-00-700-816-H01.
  - (a) If the MAT shows NOT ACTIVE for the maintenance message (or if the message does not show), then you corrected the fault.
  - (b) If the MAT shows ACTIVE for the maintenance message, then continue with this fault isolation procedure at the subsequent step.

#### ----- END OF TASK -----

#### 812. Left Engine Ignition System #1 (Upper) Power Unavailable - Fault Isolation

#### A. Maintenance Messages

(1) This task is for maintenance message: 74-33941.

#### B. Description

(1) Electrical power to left engine ignition exciter #1 (upper) is not available to the EEC when the fuel control switch is in the RUN position. This fault can be reported whenever the EEC is powered.

#### C. Initial Evaluation

- (1) Put the left engine FUEL CONTROL switch, on the P10 control stand, to the RUN position.
- (2) Set the left engine EEC MAINT POWER switch on the overhead maintenance panel, P61, to the TEST position.
  - (a) If there is a fault, then the MAT will show the maintenance message in 120 seconds or less.
  - (b) If the MAT shows ACTIVE for the maintenance message, then do the fault isolation procedure below.

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(c) If the MAT shows NOT ACTIVE for the maintenance message, then there was an intermittent fault.

NOTE: If you have an intermittent fault, you must use your judgment (and your airlines policy) to make a decision if you will replace components and if so, which components to replace. Then monitor the airplane on the subsequent flight.

#### D. Fault Isolation Procedure



USE EXTREME CAUTION WHEN EXAMINING FOR ELECTRICAL POWER. HIGH VOLTAGE CAN CAUSE INJURY OR DEATH.



REMOVE ELECTRICAL POWER FROM THE EEC BEFORE YOU DISCONNECT THE ELECTRICAL CONNECTORS. IF YOU DO NOT OBEY THIS INSTRUCTION, DAMAGE TO THE EEC CAN OCCUR.

- (1) Make sure the left engine EEC MAINT POWER switch, on the overhead maintenance panel, P61, is in the NORM position.
- (2) Open these circuit breakers and install safety tags:

#### Overhead Circuit Breaker Panel, P11

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
Α	1	C74401	L ENG STBY IGN 1
Α	2	C74407	L ENG STBY IGN 2
Е	1	C74403	L ENG IGN 1
Ε	14	C74405	L ENG IGN 2

- (3) Disconnect connector DM73003A from the EEC, M73003 (WDM 74-11-11).
- (4) Remove the safety tags and close these circuit breakers:

#### **Overhead Circuit Breaker Panel, P11**

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
Α	1	C74401	L ENG STBY IGN 1
Α	2	C74407	L ENG STBY IGN 2
Е	1	C74403	L ENG IGN 1
E	14	C74405	L ENG IGN 2

- (5) Put the left FUEL CONTROL switch, on the P10 control stand, to the RUN position.
- (6) Check for 115 VAC at connector DM73003A, pins A and B.
- (7) If there is not 115 VAC at connector DM73003A, pins A and B, then do these steps:
  - (a) Disconnect connector D11032P from the left power management panel, P110, and check for 115 VAC at pin 38 with reference to airplane ground.
  - (b) If there is 115 VAC at connector D11032, pin 38, then replace the left engine igniter 1 transfer relay, K74101, and the left engine ignition/fire detection relay, K76103 at the power management panel, P110.
    - 1) Do the repair confirmation at the end of this task.

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- 2) If the problem continues, do a check of the wiring between connector D11032P, pin 37 at the left power management panel, P110, and the EEC, M73003, connector DM73003, pin A and repair the problems that you find.
- (c) If there is not 115 VAC at connector D11032P, pin 38, then do these steps:
  - Do a check of the wiring between the left power management panel, P110, connector D11032P, pin 38 and the L ENG IGN 1 circuit breaker, C74403 on the overhead circuit breaker panel, P11.
  - 2) Do a check of the wiring between the left power management panel, P110, connector D11016P, pin 30 and the L ENG STBY IGN 1 circuit breaker, C74401 on the overhead circuit breaker Panel, P11.
    - a) Do the repair confirmation at the end of this task.
- (8) If there is 115 VAC at connector DM7003A, pins A and B, then do these steps:
  - (a) Replace the ignition exciter #1, M74001.

These are the tasks:

Igniter (System 1 or System 2) Removal, AMM TASK 74-21-02-000-801-H01 Igniter (System 1 or System 2) Installation, AMM TASK 74-21-02-400-801-H01.

- 1) Do the repair confirmation at the end of this task.
- (b) If the problem continues, replace the EEC, M73003.

These are the tasks:

EEC (FADEC) Removal, AMM TASK 73-21-15-000-801-H01

EEC (FADEC) Installation, AMM TASK 73-21-15-400-801-H01.

1) Do the repair confirmation at the end of this task.

#### E. Repair Confirmation

- (1) Do this task: Test No. 17 Engine Ignition System Test (MAT Initiated Test), AMM TASK 71-00-00-700-816-H01.
  - (a) If the MAT shows NOT ACTIVE for the maintenance message (or if the message does not show), the you corrected the fault.
  - (b) If the MAT shows ACTIVE for the maintenance message, the continue with this fault isolation procedure at the subsequent step.

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#### 813. Right Engine Ignition System #1 (Upper) Power Unavailable - Fault Isolation

#### A. Maintenance Messages

(1) This task is for maintenance message: 74-33942.

#### B. Description

 Electrical power to right engine ignition exciter #1 (upper) is not available to the EEC when the fuel control switch is in the RUN position. This fault can be reported whenever the EEC is powered.

#### C. Initial Evaluation

- (1) Put the right engine FUEL CONTROL switch, on the P10 control stand, to the RUN position.
- (2) Set the right engine EEC MAINT POWER switch on the overhead maintenance panel, P61, to the TEST position.

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- (a) If there is a fault, then the MAT will show the maintenance message in 120 seconds or less.
- (b) If the MAT shows ACTIVE for the maintenance message, then do the fault isolation procedure below.
- (c) If the MAT shows NOT ACTIVE for the maintenance message, then there was an intermittent fault.

NOTE: If you have an intermittent fault, you must use your judgment (and your airlines policy) to make a decision if you will replace components and if so, which components to replace. Then monitor the airplane on the subsequent flight.

#### D. Fault Isolation Procedure



USE EXTREME CAUTION WHEN EXAMINING FOR ELECTRICAL POWER. HIGH VOLTAGE CAN CAUSE INJURY OR DEATH.



REMOVE ELECTRICAL POWER FROM THE EEC BEFORE YOU DISCONNECT THE ELECTRICAL CONNECTORS. IF YOU DO NOT OBEY THIS INSTRUCTION, DAMAGE TO THE EEC CAN OCCUR.

- (1) Make sure the right engine EEC MAINT POWER switch, on the overhead maintenance panel, P61, is in the NORM position.
- (2) Open these circuit breakers and install safety tags:

#### Overhead Circuit Breaker Panel, P11

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
Α	14	C74400	R ENG STBY IGN 1
Α	15	C74406	R ENG STBY IGN 2
Е	2	C74404	R ENG IGN 2
Е	15	C74402	R ENG IGN 1

- (3) Disconnect connector DM73003A from the EEC, M73003 (WDM 74-11-21).
- (4) Remove the safety tags and close these circuit breakers:

#### **Overhead Circuit Breaker Panel, P11**

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
Α	14	C74400	R ENG STBY IGN 1
Α	15	C74406	R ENG STBY IGN 2
E	2	C74404	R ENG IGN 2
Ε	15	C74402	R ENG IGN 1

- (5) Put the right FUEL CONTROL switch, on the P10 control stand, to the RUN position.
- (6) Check for 115 VAC at connector DM73003A, pins A and B.
- (7) If there is not 115 VAC at connector DM73003A, pins A and B, then do these steps:
  - (a) Disconnect connector D21032P from the right power management panel, P210, and check for 115 VAC at pin 58 with reference to airplane ground.

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- (b) If there is 115 VAC at connector D21032, pin 58, then replace the right engine igniter 1 transfer relay, K74201, and the right engine ignition/fire detection relay, K76203 at the power management panel, P210.
  - 1) Do the repair confirmation at the end of this task.
  - 2) If the problem continues, do a check of the wiring between connector D21032P, pin 57 at the right power management panel, P210, and the EEC, M73003, connector DM73003, pin A and repair the problems that you find.
- (c) If there is not 115 VAC at connector D21032P, pin 58, then do these steps:
  - 1) Do a check of the wiring between the right power management panel, P210, connector D21032P, pin 58 and the R ENG IGN 1 circuit breaker, C74402 on the overhead circuit breaker panel, P11.
  - 2) Do a check of the wiring between the right power management panel, P210, connector D21016P, pin 22 and the R ENG STBY IGN 1 circuit breaker, C74400 on the overhead circuit breaker Pnel, P11.
    - a) Do the repair confirmation at the end of this task.
- (8) If there is 115 VAC at connector DM7003A, pins A and B, then do these steps:
  - (a) Replace the ignition exciter #1, M74001.

These are the tasks:

Igniter (System 1 or System 2) Removal, AMM TASK 74-21-02-000-801-H01 Igniter (System 1 or System 2) Installation, AMM TASK 74-21-02-400-801-H01.

- 1) Do the repair confirmation at the end of this task.
- (b) If the problem continues, replace the EEC, M73003.

These are the tasks:

EEC (FADEC) Removal, AMM TASK 73-21-15-000-801-H01 EEC (FADEC) Installation, AMM TASK 73-21-15-400-801-H01.

1) Do the repair confirmation at the end of this task.

#### E. Repair Confirmation

- (1) Do this task: Test No. 17 Engine Ignition System Test (MAT Initiated Test), AMM TASK 71-00-00-700-816-H01.
  - (a) If the MAT shows NOT ACTIVE for the maintenance message (or if the message does not show), the you corrected the fault.
  - (b) If the MAT shows ACTIVE for the maintenance message, the continue with this fault isolation procedure at the subsequent step.

	<b>END</b>	OF	<b>TASK</b>	
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#### 815. Left Engine Ignition System #2 (Lower) Power Unavailable - Fault Isolation

#### A. Maintenance Messages

(1) This task is for maintenance message: 74-33951.

#### B. Description

(1) Electrical power to left engine ignition exciter #2 (lower) is not available to the EEC when the fuel control switch is in the RUN position. This fault can be reported whenever the EEC is powered.

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#### C. Initial Evaluation

- (1) Put the left engine FUEL CONTROL switch, on the P10 control stand, to the RUN position.
- (2) Set the left engine EEC MAINT POWER switch on the overhead maintenance panel, P61, to the TEST position.
  - (a) If there is a fault, then the MAT will show the maintenance message in 120 seconds or less.
  - (b) If the MAT shows ACTIVE for the maintenance message, then do the fault isolation procedure below.
  - (c) If the MAT shows NOT ACTIVE for the maintenance message, then there was an intermittent fault.

NOTE: If you have an intermittent fault, you must use your judgment (and your airlines policy) to make a decision if you will replace components and if so, which components to replace. Then monitor the airplane on the subsequent flight.

#### D. Fault Isolation Procedure



USE EXTREME CAUTION WHEN EXAMINING FOR ELECTRICAL POWER. HIGH VOLTAGE CAN CAUSE INJURY OR DEATH.



REMOVE ELECTRICAL POWER FROM THE EEC BEFORE YOU DISCONNECT THE ELECTRICAL CONNECTORS. IF YOU DO NOT OBEY THIS INSTRUCTION, DAMAGE TO THE EEC CAN OCCUR.

- (1) Make sure the left engine EEC MAINT POWER switch, on the overhead maintenance panel, P61, is in the NORM position.
- (2) Open these circuit breakers and install safety tags:

#### Overhead Circuit Breaker Panel, P11

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
Α	1	C74401	L ENG STBY IGN 1
Α	2	C74407	L ENG STBY IGN 2
Е	1	C74403	L ENG IGN 1
Е	14	C74405	L ENG IGN 2

- (3) Disconnect connector DM73003B from the EEC, M73003 (WDM 74-11-11).
- (4) Remove the safety tags and close these circuit breakers:

#### **Overhead Circuit Breaker Panel, P11**

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
Α	1	C74401	L ENG STBY IGN 1
Α	2	C74407	L ENG STBY IGN 2
Ε	1	C74403	L ENG IGN 1
Ε	14	C74405	L ENG IGN 2

- (5) Put the left FUEL CONTROL switch, on the P10 control stand, to the RUN position.
- (6) Check for 115 VAC at connector DM73003B, pins A and B.
- (7) If there is not 115 VAC at connector DM73003B, pins A and B, then do these steps:

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- (a) Disconnect connector D11042P from the left power management panel, P110, and check for 115 VAC at pin 12 with reference to airplane ground.
- (b) If there is 115 VAC at connector D11042, pin 12, then replace the left engine igniter 2 transfer relay, K74102, and the left engine generator control/ignition relay, K76104 at the power management panel, P110.
  - 1) Do the repair confirmation at the end of this task.
  - 2) If the problem continues, do a check of the wiring between connector D11039P, pin 12 at the left power management panel, P110, and the EEC, M73003, connector DM73003B, pin A and repair the problems that you find.
- (c) If there is not 115 VAC at connector D11042P, pin 12, then do these steps:
  - 1) Do a check of the wiring between the left power management panel, P110, connector D11042P, pin 12 and the L ENG IGN 2 circuit breaker, C74405 on the overhead circuit breaker panel, P11.
  - Do a check of the wiring between the left power management panel, P110, connector D11016P, pin 31 and the L ENG STBY IGN 2 circuit breaker, C74407 on the overhead circuit breaker Panel, P11.
    - a) Do the repair confirmation at the end of this task.
- (8) If there is 115 VAC at connector DM7003B, pins A and B, then do these steps:
  - (a) Replace the ignition exciter #2, M74002.

These are the tasks:

Ignition Exciter (System 1 or System 2) Removal, AMM TASK 74-11-01-000-801-H01 Ignition Exciter (System 1 or System 2) Installation, AMM TASK 74-11-01-400-801-H01.

- 1) Do the repair confirmation at the end of this task.
- (b) If the problem continues, replace the EEC, M73003.

These are the tasks:

EEC (FADEC) Removal, AMM TASK 73-21-15-000-801-H01

EEC (FADEC) Installation, AMM TASK 73-21-15-400-801-H01.

1) Do the repair confirmation at the end of this task.

#### E. Repair Confirmation

- (1) Do this task: Test No. 17 Engine Ignition System Test (MAT Initiated Test), AMM TASK 71-00-00-700-816-H01.
  - (a) If the MAT shows NOT ACTIVE for the maintenance message (or if the message does not show), the you corrected the fault.
  - (b) If the MAT shows ACTIVE for the maintenance message, the continue with this fault isolation procedure at the subsequent step.

	Е	ND	OF	<b>TASK</b>	. —
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#### 816. Right Engine Ignition System #2 (Lower) Power Unavailable - Fault Isolation

#### A. Maintenance Messages

(1) This task is for maintenance message: 74-33952.

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#### B. Description

(1) Electrical power to right engine ignition exciter #2 (lower) is not available to the EEC when the fuel control switch is in the RUN position. This fault can be reported whenever the EEC is powered.

#### C. Initial Evaluation

- (1) Put the right engine FUEL CONTROL switch, on the P10 control stand, to the RUN position.
- (2) Set the right engine EEC MAINT POWER switch on the overhead maintenance panel, P61, to the TEST position.
  - (a) If there is a fault, then the MAT will show the maintenance message in 120 seconds or less.
  - (b) If the MAT shows ACTIVE for the maintenance message, then do the fault isolation procedure below.
  - (c) If the MAT shows NOT ACTIVE for the maintenance message, then there was an intermittent fault.

NOTE: If you have an intermittent fault, you must use your judgment (and your airlines policy) to make a decision if you will replace components and if so, which components to replace. Then monitor the airplane on the subsequent flight.

#### D. Fault Isolation Procedure



USE EXTREME CAUTION WHEN EXAMINING FOR ELECTRICAL POWER. HIGH VOLTAGE CAN CAUSE INJURY OR DEATH.



REMOVE ELECTRICAL POWER FROM THE EEC BEFORE YOU DISCONNECT THE ELECTRICAL CONNECTORS. IF YOU DO NOT OBEY THIS INSTRUCTION, DAMAGE TO THE EEC CAN OCCUR.

- (1) Make sure the right engine EEC MAINT POWER switch, on the overhead maintenance panel, P61, is in the NORM position.
- (2) Open these circuit breakers and install safety tags:

#### **Overhead Circuit Breaker Panel, P11**

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
Α	14	C74400	R ENG STBY IGN 1
Α	15	C74406	R ENG STBY IGN 2
Е	2	C74404	R ENG IGN 2
Ε	15	C74402	R ENG IGN 1

- (3) Disconnect connector DM73003B from the EEC, M73003 (WDM 74-11-21).
- (4) Remove the safety tags and close these circuit breakers:

#### **Overhead Circuit Breaker Panel, P11**

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
Α	14	C74400	R ENG STBY IGN 1
Α	15	C74406	R ENG STBY IGN 2
Ε	2	C74404	R ENG IGN 2

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(Continued)

Overhead Circuit Breaker Panel, P11

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
Е	15	C74402	R ENG IGN 1

- (5) Put the right FUEL CONTROL switch, on the P10 control stand, to the RUN position.
- (6) Check for 115 VAC at connector DM73003B, pins A and B.
- (7) If there is not 115 VAC at connector DM73003B, pins A and B, then do these steps:
  - (a) Disconnect connector D21042P from the right power management panel, P210, and check for 115 VAC at pin 24 with reference to airplane ground.
  - (b) If there is 115 VAC at connector D21042, pin 24, then replace the right engine igniter 2 transfer relay, K74202, and the right engine generator control/ignition relay, K76204 at the power management panel, P210.
    - 1) Do the repair confirmation at the end of this task.
    - If the problem continues, do a check of the wiring between connector D21039P, pin 23 at the right power management panel, P210, and the EEC, M73003, connector DM73003B, pin A and repair the problems that you find.
  - (c) If there is not 115 VAC at connector D21042P, pin 24, then do these steps:
    - Do a check of the wiring between the right power management panel, P210, connector D21042P, pin 24 and the R ENG IGN 2 circuit breaker, C74404 on the overhead circuit breaker panel, P11.
    - 2) Do a check of the wiring between the right power management panel, P210, connector D21046P, pin 24 and the R ENG STBY IGN 2 circuit breaker, C74406 on the overhead circuit breaker Panel, P11.
      - a) Do the repair confirmation at the end of this task.
- (8) If there is 115 VAC at connector DM7003B, pins A and B, then do these steps:
  - (a) Replace the ignition exciter #2, M74002.

These are the tasks:

Ignition Exciter (System 1 or System 2) Removal, AMM TASK 74-11-01-000-801-H01 Ignition Exciter (System 1 or System 2) Installation, AMM TASK 74-11-01-400-801-H01.

- 1) Do the repair confirmation at the end of this task.
- (b) If the problem continues, replace the EEC, M73003.

These are the tasks:

EEC (FADEC) Removal, AMM TASK 73-21-15-000-801-H01

EEC (FADEC) Installation, AMM TASK 73-21-15-400-801-H01.

1) Do the repair confirmation at the end of this task.

#### E. Repair Confirmation

- (1) Do this task: Test No. 17 Engine Ignition System Test (MAT Initiated Test), AMM TASK 71-00-00-700-816-H01.
  - (a) If the MAT shows NOT ACTIVE for the maintenance message (or if the message does not show), the you corrected the fault.

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(b) If the MAT shows ACTIVE for the maintenance message, the continue with this fault isolation procedure at the subsequent step.

----- END OF TASK -----

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#### 802. EICAS Message Latched by AIMS - Fault Isolation

#### A. Initial Evaluation

NOTE: AIMS can latch this EICAS message when it occurs. After you find the cause of the fault and correct it, it is possible that the EICAS message will continue to show.

- (1) Do not erase the EICAS message until you complete the task for the correlated maintenance message.
- (2) When this EICAS message occurs, do these steps:
  - (a) Make sure that you have the correlated maintenance message number that shows on the MAT with the EICAS message.
  - (b) Go back to the FIM Fault Code Index and find the fault code for the EICAS message.
  - (c) Find the correlated maintenance message number and the task number to the right of the fault code.
  - (d) Go to the specified task in the FIM and do the steps in the task.
  - (e) After you do the actions in the task to correct the fault, do these steps:
    - 1) Look at the MAT for the EICAS message.
    - 2) If the MAT shows LATCHED for the EICAS message, then you must erase it from the EICAS status display (AMM TASK 31-61-00-800-802).

----- END OF TASK -----

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